

Errata

Unsteady Pressures and Forces During Transonic Buffeting of a Supercritical Airfoil

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[J Aircraft, 21, 439-441(1984)]

THE caption that appears beneath Fig 4 in the above paper is actually a duplicate of the caption that appears beneath Fig 3 in the paper by J Katz and D Levin that follows. The correct caption should read: Fig 4 Steady and unsteady pressure coefficients along airfoil chord for $M_\infty = 0.805$, $C_L = 0.727$, $\alpha = 3.5^\circ$, and $Re = 21 \times 10^6$.

Two additional errors appear in the text. On page 439 in the third paragraph under the heading "Test Facilities and Model," ($0.501 \leq M_\infty < 0.805$) should read ($0.501 \leq M_\infty \leq 0.805$) and ($15 < Re \times 10^{-6} \leq 21$) should read ($15 \leq Re \times 10^{-6} \leq 21$).

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Effects of Compressor Hub Treatment on Stator Stall and Pressure Rise

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IN the above paper, reference is made to Table 1. That table inadvertently omitted from the final published version of the article appears below

Table 1 Stator geometry

	Low stagger	High stagger
Midspan stagger angle (measured from axial) deg	22.5	42.5
Camber deg	30.0	30.0
Solidity at midspan	1.0	1.0
Blade clearance mm	0.8	1.5
Number of blades	45	45
Chord, mm	38.0	38.0

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